GLOBAL ORBIT FEEDBACK UTILIZING ANALOG AND DIGITAL TECHNOLOGIES

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Abstract

At the NSLS, an analog global orbit feedback system is used in regular operations, and a digital global orbit feedback system is available in machine physics studies on the X-Ray Ring. Here, we discuss the relative merits of utilizing analog and digital technology in orbit feedback. Results of experiments are reported characterizing the performance of our analog and digital systems when operated individually or together. We give our thoughts on plans for future development of the orbit feedback systems at the NSLS.

1 INTRODUCTION

The analog global orbit feedback systems have been operational on the NSLS storage rings since 1991. These are based on correcting only a few harmonics near the tune [1] and they utilize only one-third of the available pues and trims. The analog systems provide an order of magnitude reduction in orbit motion using inexpensive hardware. The analog feedback controller is wideband and provides a stable feedback system with orbit correction bandwidth of 200 hz [2, 3]. From dc to 2 hz, the harmonic content of the orbit is corrected by a factor of 100, but at higher frequencies, the correction reduces linearly with frequency reaching to unity at 200 hz. We are also developing a digital orbit feedback system [4, 5] which has been used during machine physics studies in the X-ray storage ring. It employes up to 25 eigenvectors and all available pues and trims. The digital feedback system provides a significant improvement in slow orbit variation (less than 1 Hz), however, its correction bandwidth of 20 hz provides only moderate improvement for fast orbit motion.

We have three basic modes of operating the orbit feedback systems: (1) Run only the analog orbit feedback system as is the case during regular operation. (2) Run only digital feedback system, presently only in machine studies. In this case, the digital feedback system uses up to 25 eigenvectors and is highly effective in correcting orbit drift. The eigenvalues of the response matrix (trim excitation to pue response), with no feedback, are given in Table 1. (3) First the analog feedback is turned on, and then the digital feedback is turned on. In this mode, the digital feedback system uses a modified response matrix obtained with the analog system on. Using about 10 eigenvectors, the combined system provides additional orbit correction beyond that provided by the analog system alone. The eigenvalues of the response matrix, with analog feedback on, are given in Table 2. Fig. 1 and Fig. 2 provide improvement results for horizontal orbit motion in the slow drift and in fast orbit oscillation, respectively. The fast orbit oscillations are due to the NSLS booster, cycling at about 0.7 Hz.



Figure 1: Horizontal orbit slow drift with (a) No feedback, (b) Analog feedback only and (c) Analog and Digital feedbacks.



Figure 2: Horizontal orbit fast motion due to cycling NSLS Booster with (a) No feedback, (b) Analog feedback, and (c) Analog and Digital feedbacks.

We have observed that the vacuum chamber moves when it is heated by synchrotron radiation [3]. In the horizontal plane, the amount of this chamber motion is large at some pue locations, resulting in significant orbit measurement errors. This motion must be taken into account in order to use the digital system to improve the horizontal orbit. During studies, when the digital feedback is on, we observe that the photon beam monitored at the diagnostic beamline X28 shows a larger horizontal beam excursion with digital feedback on, although motion appears to be smaller on the PUEs (see Fig. 1). We are in process of studying the chamber motion and plan to incorporate a correction for PUE motion in order to implement the digital feedback system effectively.

2 ANALOG VS DIGITAL

The set up of our analog and digital orbit feedback systems is shown in Fig. 3. The plant P represents the trim to pue response, the controller C_a is the analog feedback element and the controller C_d is the digital feedback element. The digital feedback controller C_d has two configurations, one without analog feedback and other with analog feedback.

If the background orbit motion is represented by N as shown in Fig. 3, with both analog and digital feedbacks off, the orbit is given by the uncorrected background orbit motion N. With feedbacks on, it can be easily shown that the residual orbit is $[N/(1+P(C_a+C_d))]$ and the performance of a feedback depends on the value of $(1 + P(C_a + C_d))$. At dc, this expression has a value of 100 and the noise reduction will be by two order of magnitude.



Figure 3: NSLS X-ray ring orbit feedback systems.

Let us comment upon some of the relative merits of analog and digital feedback, based upon our experience, which is constrained by existing hardware. The Bittner-Biscardi receivers provide a very precise, low noise analog output, with a bandwidth of larger than 2 khz. The analog feedback systems we have built utilizing the output of the receivers achieve high resolution and wide bandwidth in a straightforward and economical manner. For our relatively small storage rings, utilizing 16 (X-ray ring) or 8 (VUV ring) pues and trims is easy, and provides for a significant improvement in orbit stability. One downside of the analog system is the restriced flexibility in the choice of pues and trims, which limits studies of the dependence of system performance on the design choices of pues, trims and filters. For large rings, requiring more pues and trims, the analog approach may become less attractive.

In our implementation of a digital system, the achievement of submicron resolution, low noise, and high bandwidth has been more difficult than in the analog system. The noise level in the analog to digital converter (ADC) hardware results in a measurement error of a few microns.

The bandwidth of the digital controller is presently limited in our case due to ADC sampling limit of 550 Hz, and the speed of the cpu processor used to compute and implement the orbit correction. However, these difficulties may be less restricting when designing a new system, and also will be eased by the rapid progress being made in digital components. On the other hand, the great flexibility made possible by the digital approach has already provided us with important benefits. We can easily add or remove pues or trims from the feedback, to assess their effect. Also, we can easily choose from different orbit correction algorithms, and vary the number of eigenvectors included in the correction. In this way, we have achieved new insight into the optimum conditions for feedback, which has even allowed us to improve the analog vertical feedback on the X-ray ring (see Fig. 4). In studies of digital feedback it was noticed that there are 8 vertical eigenvalues of magnitude greater than unity (see table 1), while existing analog feedback was only using 8 PUES and 6 eigenvectors. Based on this observation, we modified the analog vertical orbit feedback to include 16 PUES and 8 eigenvectors, improving the orbit correction by about a factor of 3. The improvement results are shown in Fig. 4.



Figure 4: X-Ray vertical orbit drift with (a) No feedback, (b) Analog feedback with 8 PUEs and 6 eigenvectors, and (c) Analog feedback with 16 PUEs and 8 eigenvectors.

3 FUTURE PLANS

In short term, we plan to study the performance of digital feedback system by upgrading it with a) new ADC orbit measurement system which will have lower noise level by an order of magnitude, and will sample at a higher rate of about 1 Khz; and b) new cpu processor which is faster by a factor of 3. Simulation of feedback system has shown that these improvements should increase the digital feedback bandwidth from 20 hz to 40 hz. We will, also, study the improvement resulting from incorporating correction of pue motion due to vacuum chamber movement.

In long term, we will look into an orbit correction system which utilizes digital feedback only. We expect this sys-

Eigen	Horizontal	Vertical
number		
01	17.62	26.88
02	11.23	16.36
03	3.07	4.89
04	2.97	4.65
05	2.09	3.68
06	1.68	3.44
07	1.64	3.01
08	1.53	2.58
09	1.19	.30
10	.98	.29
11	.95	.29
12	.72	.28
13	.70	.28
14	.63	.28
15	.62	.28
16	.45	.23
17	.34	.13
18	.28	.13
19	.27	.12
20	.26	.12
21	_	-
22	_	-

Eigen	Horizontal	Vertical
number		
01	1.34	.30
02	1.27	.29
03	1.06	.29
04	.89	.29
05	.74	.28
06	.71	.28
07	.62	.28
08	.59	.23
09	.48	.20
10	.40	.14
11	.32	.13
12	.30	.13
13	.28	.12
14	.28	.12
15	.27	.12
16	.24	.11
17	.23	.11
18	.20	.11
19	.05	.10
20	.05	.10
21	—	-
22	_	-

Table 1: Eigenvalues of horizontal and vertical response matrix with no feedback. For vertical, note that there are 8 eigenvalues of magnitude greater than unity.

tem will sample and process orbit data at a rate of several khz. Also, this system should integrate with an orbit measurement system which incorporates correction for PUE's motion.

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5 REFERENCES

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