Abstract

When complete, the Continuous Electron Beam Accelerator Facility will house a 4 GeV recirculating linear accelerator containing 42/4 cryomodules arrayed in two antiparallel lines and an injector. Currently, over half of the accelerator containing 42/4 cryomodules have been installed. Each cryomodule contains eight superconducting niobium 5-cell rf cavities that operate at 1.497 GHz [1]. A cryomodule must provide an energy gain of 20 MeV to the 200 μA beam [2]. The resultant dynamic heatload must be less than 45 W. The cavity parameters that are measured during the commissioning process include the external Q's of the cavity ports, the unloaded Q (QL) of the cavity as a function of accelerating gradient, and the maximum operating gradient of the cavity. The sensitivity of the resonant frequency to changes in pressure and gradient is also measured. Finally, the mechanical tuners are cycled and characterized. In all cases, the performance of CEBAF cryomodules has exceeded the design requirements. A portable test stand that is controlled by a Macintosh computer. This system is capable of fully interlocked local control of the rf system. With this system, the commissioning of a cryomodule can be completed in about three eight-hour shifts.

II. DESCRIPTION OF THE TEST SYSTEM

A block diagram of the test system is shown in Figure 1. A voltage controlled oscillator (VCO) is used to control the klystron amplifier. The VCO can operate in either a continuous wave (CW) mode or a gated pulsed mode. The pulsed mode is used to measure the emitted power (Pe) from the cavity and to measure the loaded Q (QL). A PIN diode is used as a gating device. The VCO uses a phase-lock loop to track the cavity field probe signal. The VCO is connected to a fast shutdown (FSD) node so that rf will be turned off by any fault signal in the interlock chain. These faults include waveguide window arc and temperature faults. The rf will also be turned off by waveguide and beamline vacuum faults, liquid level, and helium pressure faults through the fast shutdown node. Once the klystron drive signal has been tuned and the amplitude has been set, the computer can control the pulsed/CW mode of the VCO.

The VCO routes the gated signal, Pe, to an analog power meter, and to the spectrum analyzer. The analog meter integrates the measurement of the pulsed Pe signal over time. The spectrum analyzer is used in the time domain mode to measure the decay time of the emitted power signal.

Three digital power meter channels are available to measure the incident power (Pi) from the klystron, the transmitted power (Pt) from the cavity field probe, and the reflected power level behind the circulator (Pm). This last signal is part of the interlock chain and is necessary to protect the klystron. A 18 GHz frequency counter is available for fast frequency measurement. All the instruments other than the analog power meter have GPIB capability.

A heater controller is used to power one of the four heaters that are located in the helium vessels. Heater control is required in order to make a calorimetric QL measurement. The heater power level can either be set manually or controlled by the computer.

A self-excited loop is provided that allows the klystron drive to track the resonant frequency of the cavity during tuning. The SEL is also connected to the FSD node for interlocked rf operation.

Geiger-Mueller tubes are positioned near each waveguide assembly of each cryomodule and on the beam tube at each end of the cryomodule.

A Macintosh computer is used for control of the test system, data acquisition, and measurement algorithms. The "Labview" software is an object-oriented, icon-based programming language that allows the computer to control ana-

Table 1. Cavity parameters

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Qes (fundamental power coupler)</td>
<td>6.6 x 10^6 ± 20%</td>
</tr>
<tr>
<td>Qpr (field probe)</td>
<td>1.3 x 10^11 ±65%</td>
</tr>
<tr>
<td>QL</td>
<td>≥ 2.4 x 10^9</td>
</tr>
<tr>
<td>f0 (π mode, T=2 K)</td>
<td>1.497 GHz</td>
</tr>
<tr>
<td>E_max</td>
<td>≥ 5.0 MV/m</td>
</tr>
<tr>
<td>Pressure Sensitivity</td>
<td>&lt; 60 Hz/torr</td>
</tr>
</tbody>
</table>

These parameters are all measured during the commissioning process. A QL vs. E_max curve is generated, and the effect of the ponderomotive force on the resonant frequency of the cavity is also measured. Finally the mechanical tuners are cycled several times to insure proper operation.

*This work was supported by DOE contract DE-AC05-84ER40150
log and digital I/O. This software also allows the computer to function as a GPIB controller. The computer, in its current configuration, can handle 14 analog output channels, 8 differential analog input channels, and 16 digital I/O lines.

III. MEASUREMENT SEQUENCE

Before any high-power operations begin, a series of measurements is made using a network analyzer. These measurements include a bandwidth measurement of $Q_L$ for each cavity:

$$Q_L = \frac{f_0}{\Delta f}$$  \hspace{1cm} (1)

The frequency and amplitude of each of the fundamental passband modes are measured for each cavity. Finally, a reflection measurement is made to determine attenuation from the field probe to the external port of the cryomodule for each cavity. This last measurement is combined with other attenuation measurements for cables and couplers to create a matrix of attenuation values. These are entered into the computer, allowing automatic calculation of the absolute power levels at the cavity plane. These measurements may be performed while the cavity temperature is at 4.2 K.

A complete interlock checkout is performed prior to any klystron operations. Simulated fault signals are applied to the various detector circuits while the operator observes the FSD.

When the interlock checkout and low-power measurements have been completed, and the cavity temperature has been lowered to 2.15 K or less, high-power commissioning can begin. The rf phase and amplitude control modules are disconnected from the rf system and replaced by the VCO and power meters.

The VCO drive signal is first tuned until the resonant frequency of the cavity is located. Then, the klystron power level is raised until the accelerating gradient is at about 4 MV/m. At this time, a measurement of $Q_{ex}$ is made. This is accomplished by using the fact that for a strongly overcoupled cavity ($\beta >> 1$):

$$Q_e = Q_{ex} << Q_L, Q_o$$  \hspace{1cm} (2)

$Q_e$ can then be measured by the time decay method. While the klystron is in the pulsed mode, the spectrum analyzer is used to examine the emitted power waveform. A measurement is made of the time required for a 20 dB decay in this signal. The following equation is used to calculate $Q_e$ and from that $Q_{ex}$:

$$Q_e = 4.340 \frac{\Delta f}{20 \text{ dB}}$$  \hspace{1cm} (3)

The VCO is switched to the CW mode. The incident and transmitted power levels are then acquired, allowing calculation of the accelerating gradient.

Accelerating gradient calculations begin with the following:

$$\omega U = P_o P_{ex} = P_o P_{ex} = P_o P_{ex}$$  \hspace{1cm} (4)

The gradient is proportional to the square root of the stored energy in the cavity:

$$E_{ex} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}} = \sqrt{\frac{P_o}{L}}$$  \hspace{1cm} (5)

where $\frac{k_0}{L}$ is the shunt impedance per meter, 1920 $\Omega$, and $PRF$
is the pulse rate frequency, 50 Hz. In terms of the integrated
emitted power \( P_e \):

\[
E_{\text{em}} = 8.5 \times 10^4 \sqrt{P_e} \quad (f_e = 1.497 \text{ GHz, PRF} = 50 \text{ Hz})
\] (6)

In terms of the transmitted power:

\[
E_{\text{em}} = \kappa \sqrt{P_e}
\] (7)

From equation (5) it can be seen that:

\[
\kappa \frac{Q_b}{L} = \sqrt{\frac{\beta}{L} \omega U} = \sqrt{\frac{\beta}{L} \frac{P_e}{PRF \cdot P}} = \sqrt{\frac{P_e}{P'}}.
\] (8)

\( Q_b \) can be determined from \( \kappa \):

\[
Q_b = \frac{\kappa^2}{1920}
\] (9)

In terms of the incident power, the following relation is needed

\[
P_s Q_0 = \frac{4\beta}{(1+\beta)} L Q_i
\] (10)

From equations (5) and (10), it can be seen that

\[
E_{\text{em}} = \sqrt{\frac{\beta}{L} 4\frac{P_e}{P'} \frac{Q_b}{L} L (1+\beta)} = \sqrt{\frac{4}{1+\beta} P_s Q_i} \quad (\beta >> 1)
\] (11)

Once the gradient level has been established, \( Q_0 \) can be measured. The cryomodule must be isolated from the cryogenic system by closing the supply and return valves. The change in pressure due to the static heat load \( \Delta P_s \) over some constant time (usually one minute) is measured. Then, a known power level \( H_s \) is applied to one of the internal heaters. The pressure rise due to the heater power \( \Delta P_h \) is measured. Finally, the rf is turned on at the desired field level. While the klystron is in the CW mode, the pressure rise due to the rf \( \Delta P_{\text{rf}} \) is measured. The rf heat load can then be calculated from the following equation:

\[
P_s = H_s \frac{\Delta P_{\text{RF}} - \Delta P_h}{\Delta P_{\text{rf}}}.
\] (12)

From equations (4) and (5) it can be seen that:

\[
Q_0 = \frac{E_{\text{em}}}{1920 P_s}
\] (13)

The level of the accelerating gradient is raised in 1 MV/m steps and the measurements cycle is repeated. The measurements sequence is complete when \( E_{\text{max}} \) is reached. Limiting factors for \( E_{\text{em}} \) include cavity quench, 1 W of field emission, R/hr of radiation measured at the beamline or waveguide assembly, excessive waveguide arc, window temperature, or vacuum faults that inhibit sustained operation of the cavity. A \( Q_0 \) vs. \( E_{\text{em}} \) curve is shown in Figure 2.

Once the phase and amplitude have been adjusted, the entire measurement sequence described above can be performed by the computer with about four keystrokes at each gradient level. About an hour is required to generate a \( Q_0 \) vs. \( E_{\text{em}} \) curve.

\begin{figure}[h]
\centering
\includegraphics[width=\textwidth]{figure2}
\caption{\( Q_0 \) vs. \( E_{\text{em}} \) curve.}
\end{figure}

Frequency sensitivity to pressure is measured by setting a constant accelerating gradient. The resonant frequency is then monitored while the helium pressure in the cryomodule is cycled. The effects of the ponderomotive force on the resonant frequency are measured by monitoring changes in the frequency while varying the accelerating gradient. A constant pressure should be maintained during this measurement.

Finally, the tuners are cycled. The SEL is substituted for the VCO as klystron drive. The SEL will track the resonant frequency through the full range of tuner movement. Generally, the tuners are moved through a frequency range of \( \pm 50 \) kHz around the design frequency, 1497 MHz.

IV. CONCLUSION

The commissioning test system allows fast and flexible testing of cryomodules. The use of a computer to control the test procedures minimizes the time required to test and allows most of the data reduction to be performed automatically. About half of the cryomodules have been installed and tested. Commissioning has shown that all of these cryomodules exceed the design performance requirements.

V. REFERENCES

