

SPEAR3 Commissioning J. Safranek for the SPEAR3 commissioning team

- SPEAR built 1972 e+/e- collider at Stanford/SLAC
- O 1989 became dedicated light source
- **O 2003 SPEAR3 installation**
 - Complete rebuild, maintaining geometry for photon beamlines
 - \mathbf{U} Low emittance optics, $\mathbf{\epsilon}_{\mathbf{x}}$ = 18 nm

 - **4 11 beamlines; 7 insertion devices**



Commissioning milestones

o 2003:

- **o 1 April, SPEAR2 removal begins**
- **o 9 December, Transport line commissioning begins**
- **o 10 December, First beam to SPEAR**
- **o** 15 December, First accumulation
- **o 2004:**
- o 22 January, 100 mA stored
- **o 8 March, First photons**
- **o** 15 March, Start of operations







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Commissioning Team



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Matlab Software Tools



Matlab Toolboxes for Accelerator Simulation and Control:

- AT Accelerator Simulation
- Middle Layer Software Accelerator Control + Physics Functions
- MCA Matlab to EPICS Library
- LOCO Accelerator Calibration



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Electronic logbook



- Web-based, interactive browser
- Accepts graphics and text
- o Searchable
- o Backed up
- Authors: Kay Rehlich, Raimund Kammering, DESY
- Courtesy: Patrick Krejcik, SLAC

SPEAR3 Logboo	ok - Microsoft Internet Explorer					
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0224,a 022,a 03 0125,a 0125,a 0125,a 0125,a 0125,a 0121,a 0220,a 020	SPEAR3 Logbook 25. January 2004 Solution 25.01.2004 23:33 Portmann Vertical Vacuum Chamber Scan Bumps used for a vertical scan. I'm not sure why scaling the correctors produces an angle has you go to larger bumps. It looks more like a nonlinearities in the BPMs (pincushion, etc) then actual beam motion. If it was an optics nonlinearity I would have expected more leakage. Image: Solution of the state of the scale of the sca					
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Diagnostics



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BPM IF frequency issue



○ BPM troubles –

- Fill-to-fill orbit not reproducible
- **Solution** Second Secon
- Orbit dependence on RF phase
- Solution: unlock IF
 from RF frequency





Find rf frequency that centers average orbit in sextupoles.

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RF frequency, 60 days

- RF frequency included in orbit correction/feedback
- In March temperature up, RF frequency down.



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|y| < 2 mm, |x| < 4 mm



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Beam-based alignment repeatability



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SPEAR orbit stability



• Orbit stability specs.:

- % ~3-5 μm vertical
- « ~16-40 μm horizontal
- Girder frequencies ~20 Hz
- O Invar struts supporting BPM:
- Most synchrotron radiation hits H2O-cooled masks
- Tight power supply stability specs.
- **O Bergoz BPM electronics**
 - Solution States Stat
 - Solution States Stat
- ο Alignment < 150 μm

Table 3.23 RMS source point beam dimensions and stability requirementsfor SPEAR 3 (rms, 1% coupling).

	n k	Electron	Photon	10%
ID-Wiggler Dipole	σ _x (μm)	160	160	16
	$\sigma_{x'}$ (µrad)	236	mrads	< mrad
	σ _y (μm)	51	51	5
	$\sigma_{y'}$ (µrad)	11	136	14
	σ _x (μm)	435	435	43
	$\sigma_{x'}$ (µrad)	43	2-20 mrad	< mrad
	σ _y (μm)	30	30	3†
	$\sigma_{y'}$ (µrad)	6	136	14
D-100 per und.	σ _x (μm)	435	435	43
	$\sigma_{x'}$ (µrad)	43	43	4
	σ _y (μm)	30	30	34
	$\sigma_{y'}$ (µrad)	6	15	1.5

†. This requirement can be relaxed to 5 μ m due to 50 μ m minimum vertical spot size achieved from present focusing mirrors.



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x(100mA)-x(50mA), BPM intensity dependence



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x(84mA)-x(1mA), BPM intensity dependence 476 MHz rf noise on ½ BPMs



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BPM fill pattern dependence x(16mA, single bunch)-x(16mA, multibunch)



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PSD Blue: orbit motion Red: BPM noise floor

BPM



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BPM PSD with v_s noise





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Fast orbit feedback



O Digital feedback

 Includes electron and photon BPMs.

○ 4 kHz cycle rate.

○ ~200 Hz bandwidth



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LOCO optics analysis code



- Calibrate/control optics using orbit response matrix
- **O Determined quadrupole gradients**
 - $bar{\beta}$ functions, η
 - \backsim Found .017 ν_y error from excess dipole focusing
- Corrected coupling
- Calibrated BPM gains, steering magnets
- Measured local chromaticity and transverse impedance



- New MATLAB version of code
 - rewritten from FORTRAN
 - linked to control system
 - linked to AT simulator
 - G. Portmann

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Linear optics calibration method

The orbit response matrix is defined as

$$\begin{bmatrix} \bar{X} \\ \bar{y} \end{bmatrix} = M \begin{bmatrix} \bar{\Theta}_{X} \\ \bar{\Theta}_{y} \end{bmatrix}$$

The parameters in a computer model of a storage ring are varied to minimize the χ^2 deviation between the model and measured orbit response matrices (M_{mod} and M_{meas}). $\chi^2 = \sum_{i,j} \frac{(M_{ij}^{\text{meas}} - M_{ij}^{\text{model}})^2}{\sigma_i^2} \equiv \sum_{k=i,j} E_k^2$

The σ_i are the measured noise levels for the BPMs; *E* is the error vector.

The χ^2 minimization is achieved by iteratively solving the linear equation

$$E_k^{new} = E_k + \frac{\partial E_k}{\partial K_l} \Delta K_l = 0$$

$$-E_k = \frac{\partial E_k}{\partial K_l} \Delta K_l$$

For the changes in the model parameters, K_p , that minimize $||E||^2 = \chi^2$.

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Coupling & η_y correction, LOCO

Minimize η_y and off-diagonal response matrix:



Lifetime, 19 mA, single bunch



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Measured dispersion





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European Particle Accelerator Conference, July 8, 2004

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Chromaticity



Nonlinear ξ:

(v_x, v_y) vs. f_{rf} agrees with model.



Local chromaticity calibrated with LOCO shows no sextupole errors:



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Dynamic aperture vs. ∆p/p

- Dynamic aperture measured with single injection kicker for varying rf frequency.
- $x_{\beta}(\Delta p/p)$ for different straight sections. (From model.)
- **o** RF acceptance



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○ Lifetime vs. V_{gap}, rf

o 8 mA, single bunch

- Source the second se
- $rightarrow au^{1/3} \sim rf$ acceptance





Dynamic aperture vs. tune

• Resonant lines:

- $\forall v_x v_y = 9$
- $\Rightarrow 3v_x + v_y = 48$
- $4v_x + v_y = 62$
- Resonances offset from tune shift with amplitude.
- * = operating tunes
 (14.19, 5.23)
- Data gathered automatically on owl shift.

5.0.71 4.0.57 5.35 5.35 B,2,53 2,2,39 1.62 1.2.25 20 3.1.48 0.3.16 19 18 5.3 3,31307 17 1.4.7 16 2,3,44 0.4.21 15 14 13 5.2 2.-1.23 0.5.26 12 11 **5.15** 5.15 3, 2,32 **2-2,38** 14.12 14.12 14.14 14.16 14.18 14.2 14.22 14.24 14.26 14.28 -1,512-2,4 14.3 10 **V**x**V**x 14.1 14.3

Maximum K3 amplitude [kV] versus tune (raw data)

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Lifetime vs. tunes



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Aperture scans

- $\odot \tau$ vs y in small gap ID
 - Poor man's scraper

Determine minimum gap for future IDs

- \backsim Smaller still for reduced β_v

Complications

- Vacuum degrades with beam bump
- Coupling, η_y degrades with beam bump
- Need scraper

Vertical beam bump in ID chamber





Low emittance optics test



- $\circ \eta$ = 10 cm in IDs
- $\odot \varepsilon_x$: 18 nm \rightarrow 12 nm
- **O Good lifetime, injection**
- **O Reserved for later upgrade**

LOCO fit η : nominal optics

low emittance optics



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Momentum compaction measurement



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SPEAR3: Longitudinal Dynamics



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α_2 measurement



$\odot |\alpha_2|$, sextupoles off >> $|\alpha_2|$, sextupoles on

O Energy aperture much reduced with sexupoles off



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Impedance, instabilities

 \odot 25 mA single bunch limit ($\xi_{x,y}$ = 1)

\odot Multibunch v_v oscillations

- Ion driven
- Becreasing as vacuum improves
- $\odot v_s$ oscillations driven by 360 Hz harmonics
- 200 mA tests ongoing
- **O Transverse impedance measurements**



*see V. Sajaev, PAC03.

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Lifetime (at 90 mA) vs. Integrated Current



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Acknowledgements



I would like to express my appreciation for all the help SSRL received with the commissioning effort from accelerator physicists worldwide. I would also like to commend the SSRL staff. The speed at which commissioning proceeded is a testimony to the fine work that went into designing and building SPEAR3. Finally, thanks to the SSRL operations group. Their experience, expertise and energy was a great help.